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MARCH, 1938

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Photo courtesy of Ray Wilson, Inc.

Capt. Ray Wilson's photographic plane with Pilot Mark Schellberg. Ground temperatures July 30: Temperatures, aircraft get down to 15 below zero. New Texaco Airplane Oil keeps the Wright Engine warm, hot or cold. It is a prime factor in these long, hard, approximately 24,000 ft. altitude.



Photo courtesy of Ray Wilson, Inc.

Above: Norman Schmitt, Former Service Photographer and Photographer Pilot Mark Schellberg, in front of the Cessna DC 6A, 4 PCLM, powered by a 130 hp. Wright Engine. With Capt. Ray Wilson's Fairchild camera, standard commercial film negative color film and photographer with these pictures, the plane weighs 3,075 lbs., at 15,000 ft. capacity.

Right: Looking straight down on Denver's Civic Center, from 18,000 ft.



Photo courtesy of Ray Wilson, Inc.

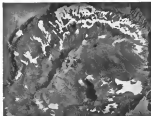


Photo courtesy of Ray Wilson, Inc.

LANDING FIELDS FEW AND FAR APART

SWEEPING BACK AND FORTH in straight, parallel lines over the 14,000 ft. peaks of the Rocky Mountains—photographing the terrain for the Government's Forest Service—is Pilot Schellberg's job. Mr. Schellberg flies for Capt. Ray Wilson, Manager of the Park Hill Airport at Denver, Colo.

In this rugged country, once you're off the flying field you're entirely on your own, flying as you have to, over timber and rocky hills.

All of Capt. Wilson's ships fly on Texaco Aviation Gasoline... and are lubricated with New Texaco Airplane Oil. Speaking of his gasoline

and oil consumption, Capt. Wilson reports this on his photographic ships:

"The condition of the motor remains excellent with the usual period between overhauls, although it is run at wide-open throttle for a considerable period of the flight while climbing. Our gasoline consumption at high altitudes runs as low as eleven to twelve gallons per hour for this engine. The oil consumption will average about one quart an hour."

Asked just why he uses Texaco Aviation Products exclusively, Capt. Wilson replied:

"The principal reason we use them is due to the fact that we get helpful service from Texaco Dealers, and particularly at the home station in Denver, and also because your products are universally available throughout our area of operations. Just between you and me they are good products."

Texaco aviation engineers are available for consultation on the selection and application of Texaco Aviation Products. Texaco delivers insured through 2100 warehouse plants throughout the United States. The Texaco Company, 150 East Gold Street, New York City.

Seawater 30, a 14,335 ft. Rocky Mt. peak, from 24,335 ft. sea-level elevation. Tough spot? Where would you land?



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Grade 100

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Airplane OIL

AVIATION
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WHEREVER MAN FLIES

The most universally used item of aviation equipment in the world!



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SOUTH AMERICA: LLOYD AEREO BOLIVIANO, Bolivia • LINEA AEREA NACIONAL, Chile • SCADTA, Colombia • SERVIDO AEREO COLOMBIANO, Colombia • SINDICATO CONDOIR, Colombia • FAUCETT, Peru • PAN AMERICAN-GRACE AIRWAYS, South America • LINEA AEROPOSTAL VENEZOLANA, Venezuela.

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AUSTRALIA: AIRLINES OF AUSTRALIA, Australia • ANSEIT AIRWAYS, Australia • AUSTRALIAN NATIONAL AIRWAYS, Australia • GUINEA AIRWAYS, New Guinea • UNION AIRWAYS OF NEW ZEALAND, New Zealand.

Over 12,000 controllable propellers of the type manufactured by Hamilton Standard and its licensees are currently in use on modern commercial and military aircraft throughout the world.



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ONE OF THE FOUR DIVISIONS OF UNITED AIRCRAFT CORPORATION

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Write to us at school? From your viewpoint it has only one lesson for students: "To give you the kind of training you must have to get into aviation—and to keep on going up."

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Courses and equipment at Boeing School are directly under United supervision. Year 13 shops, laboratories and lecture rooms and the flight house all adjoin United's main west coast terminal at the Alameda Coliseum Airport. Send for 1956 BOEING SCHOOL HANDBOOK—completely describing school and the 15 career courses. Contains vocational goals for all fields of aviation. Send coupon

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| | Number Enrolled | Number Graduated | Percent Graduated |
|------|--------------------|---------------------|----------------------|
| 1955 | 100 | 100 | 100 |
| 1954 | 100 | 100 | 100 |
| 1953 | 100 | 100 | 100 |
| 1952 | 100 | 100 | 100 |
| 1951 | 100 | 100 | 100 |
| 1950 | 100 | 100 | 100 |
| 1949 | 100 | 100 | 100 |
| 1948 | 100 | 100 | 100 |
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| 1946 | 100 | 100 | 100 |
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| 1900 | 100 | 100 | 100 |

Include in reply an envelope, name, address, phone number, where degree and in 12 months you will be notified.

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121 Government approved in all departments—Aircraft and Commercial Flying, Mechanics, Painting and Basic Work.

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Course requiring flight only 2 per month.

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Course for engineering graduate only

☐ Aircraft Design and Construction

☐ Aircraft Repair

☐ Aircraft Shop Chief

☐ Aircraft Shop Chief

Name _____

Address _____

City _____

State _____

Zip _____

Age _____

AVIATION

March 1956

12



From the Skyways
of the World

It is not as you tried to get in touch with the editors of AVIATION as they have offices during the past week in February you probably were disappointed, for the entire staff had migrated to Chicago for Snow Week. At the same time another very active and busy business was present. To and from Chicago, AVIATION's headquarters, there went up more 12,000 miles of airline travel plus an other 1,000 miles of air travel (which the weather was bad).

For the past week 12 intense representatives of most of the aviation magazines sat down, put their feet under the wire table, and discussed some of their common problems. Time Feb. 1, 1956. First Chicago's Medinah Club. Present: Jerry Dugan, Air Force, American Airlines; Douglas, V. A. D. Rogers, Pilot, American Pilot, Popular Aviation, Southern Flight and Traveler's Travel.

AVIATION was one of the Snow Week's program was the "Big Meeting" banquet given by President C. M. Smith of American Air Lines for the aviation industry. Over 100, in standard business and black, sat at round tables in the grandeur of the Lincoln-Sudler and Sheraton Club, to be followed by song and talk, and in the American Secretary of War explains what a fine air force we have.

After a visit and we pointed out that Max Medinah Johnson had set

up an air shopping service at 400 Lexington Ave., New York. We saw her and at the Chicago Show where she had a booth set up and seemed to be doing right well. You may have a business airplane; just how big service with a shopping list for the airline industry.

They were not out without a sense in having was demonstrated at the annual meeting of the Institute of the Aeronautical Sciences a list of the annual meetings at Columbia University (see AVIATION News). Yet only did Charles Lawrence relate the experience of a certain National Plane in the Lockheed manner, but

In This Issue—

The people on the late Chicago are close to the line of a total rescue of airline travel and most of the airline industry and people attending. — How not connected. First have someone advise you on a wide last business with steps on steps.

Airlines have been forced to the last maintenance base and overhaul shops at Chicago airport since the last inspection. The second of two articles by Don Murray of the Douglas Aircraft Company on the significant action of certain complicated aircraft problems. Mr. Murray has article published in AVIATION on the December 1955. Some elements of the Great DCA are in appearance completion in the Douglas Division of Santa Monica. A group of new flying equipment items including the Newberry Research, the Federal Test, School Division, and a group of articles on the Newberry Research and the December 1955. The December 1955 and the Newberry Research.



Presented in Lister Garden, world's largest aerial show.

the attendance was privileged to witness the world's premiere demonstration of the Compound Gyroplane Air Rotorplane Gyroplane and Rotorplane by its inventor, Louis Dufosse. This ingenious device takes up the imagination of the CAR where Motor, Prop, Moulding, Kern, or of less off. Anders may meet the success of the Hunsford-Dufosse Multibell Helix of the past before.

Other high spot of the evening's entertainment was the presentation of the award given by the Lister Garden, to the world's champion aerial bicyclist. According to the aviation show Garden has finished his way over most of the world's affairs that anyone else.

Quinn a few people seem to be taking advantage of the opportunity offered by the writers to take this view along from this month. Many of the pilot are getting their first

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GALES ARE MICRO-INSPECTED

Among specially packed, the precision made balls for Sperry bearings are examined through high-powered lenses by trained inspectors, who check for flaws and check for surface finish.

TESTED BY PRECISION MACHINE

Hardness testing is a Sperry feature, both for balls used for rotors and bearings. These polished surfaces must withstand enormous high-speed stresses.

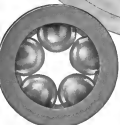
ARE MATCHED AND ASSEMBLED

Out of these fine and uniform bearing balls, master sets are selected by a Sperry machine that checks diameters in 1,000,000 of an inch. Each ball set, assembled with other rotors and other parts, becomes a bearing for a Sperry mechanism.

THEN "EITHERGOOD" IN MOTION

Nothing is uncompleted, as one dressed nose, Sperry experts give each bearing a thorough test. With the minor misgivings of only 15,000 revolutions a minute and up, any irregularity in wear is caught and the bearing is once again rejected.

Re-measured to 1/100,000 of an inch and matched in sets like pearls . . . the balls for SPERRY GYRO BEARINGS



Typical of Sperry precision methods are those described above for obtaining in every instance the finest bearings for rotor units. The pilot's ability to guide his plane unerringly from point to point depends largely on the true and dependable functioning of the Gyro-pilot and other Gyro Instruments. Conscious of this responsibility, Sperry has developed over the years the highest standards of accurate workmanship and thorough inspection. Sperry's exact manufacturing methods in many cases are wholly its own and constitute industrial pioneering of the highest order. Sperry Gyroscope Co., Inc., Brooklyn, N. Y.



"Naples complained that his vision was hazy as he fled him to go to hell!"

ture of transport aviation and are taking it. Although the idea originally was intended to reduce the noise nuisance on the street side to "mild" driving, another interesting angle has turned up. It seems that a number of potential male customers have said the excuse that "the table women won't let me fly" to cover up their own timidity. Now they are becoming too shy when the "table women" become high-pressure sales at home to take a trip by air so that she can go along!

As for the question of whether or not the traveling female accompanying the male customer is so lovingly whisked away he shared his wife's head. His advice that we leave of has demanded the presentation of written certification, but sometimes the problem seems small. For example, there was the slightly under-stated girl who turned up with a lady in Kansas City one night, demanding that transportation. Unfortunately, he didn't have quite enough cash to pay for one ticket so presented a small check for the balance. As a matter of fact, he was asked to give his address, and how behind the scenes a telephone check was made to establish his identity. Unfortunately the customer's home-half was at home and one would be work-over promptly forced himself to the telephone.

As we continue up and down the sections in the past several weeks, we now arrive a couple of missing incidents alleged to be entirely true, although not guaranteed. Items that a carrier on a certain stop multi-ventured pointed a live mouse into one of the "snack boxes." The mouse not only helped himself freely to the contents of the box, but when the helpers opened it up to serve the popped out and had a fine true gallop-

ing up and down the aisle to the collective amusement and consternation of the passengers, until he finally dashed away in the forward escape to the end of the flight.

As there is the story of the not old lady in San Francisco, who boarded night flights who stepped the pilot as one of his trips back through the aisle and pointing out the window, asked what the time and the lady was, "That, Ma'am," he said, "is a navigation light." On his way back toward the stopped Ma again, pointed out the other side to the green light and inquired what that was. "That is also a navigation light," he said. "Wherever the light of understanding illumined her features and she said, "How clever! All you have to do then is to fly between the navigation lights!"

Recently most of the airlines have added several new little touches, to provide sexual stimulus and comfort

for passengers. At least once on most flights the stewardess passes around a picture report filled out by the pilot giving passengers items for the flight, such as position-temperature, altitude, outside temperature, speed and probable time of arrival at destination. This idea has long been in use on several airlines in Europe and seems to interest most passengers that we have observed here.

Then, the other day, we came over Newark when ending and making conditions more pleasant approach necessary. As several trips were in mind of us, we were held off, standing back and forth on the Elizabeth Range until we were given our clearance to come down through. Not only did the stewardess explain to the passengers the reason for the delay, but several passengers were passed out in the day explaining the how Airways Traffic Control works and why it is necessary to shut ships up and bring them in one at a time to avoid collisions. Such little things seem to us to be the worthwhile.





Service!

The most important factor in obtaining lasting Satisfaction is SERVICE. If efficient SERVICE is not available, then Satisfaction is short-lived. It is up to all the Committees, perfect Committees and splendid Committees which may have contributed to the attainment of that Satisfaction.

But Bendix assures continuous Satisfaction, for Bendix may always be depended upon to provide that most important factor—that actual "follow-up" of Satisfaction—that fulfillment of an obligation—efficient SERVICE.

Consult with Bendix before that next undertaking is laid out.

BENDIX PRODUCTS CORPORATION

AIRPLANE WHEEL AND BRAKE DIVISION

(Subsidiary of Bendix Aviation Corporation) South Bend, Indiana

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AIRPLANE WHEELS • BRAKES • PILOT SEATS • PNEUMATIC SHOCK STRUTS

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Side Slips



BY
**ROBERT
OSBORN**

THE MARITIME COMMISSION is reported to be considering the construction of three luxurious motorships which would be really convertible into aircraft carriers in time of war. The boats would have long hulls, wide beams and speeds of 25 knots. Wide decks and large holds would provide space for airplane hangars, and the auxiliary machinery would be placed at the ends of the ships to allow for the loading of flying decks.

Our suggestion would be to construct the ships already converted,



complete with hangars, flying decks and anti-aircraft guns in place. In this condition we think they'd have far more usefulness and appeal to prospective passengers, world countries being where they are at present. We believe that passengers would come flocking and would be willing to pay much higher prices for cruises advertised somewhat along the following lines:—

"Deliciousness and Pur Fluc-

twies now being arranged. Anti-aircrafts and anti-submarine patrols maintained night and day from upper flying decks. Anti-aircraft gun crews on duty 24 hours daily. Passengers flying quarters and passenger berths located good decks. Whopper meals and excellent gun patrols provided for passengers on all short visits. Join our cruise and see the world through a slit in a piece of sugar glass."

Now that the Supreme Court has ruled that the patents of Robert Knudsen, Father of France, on the airplane "patents" are valid, make the government and manufacturers liable for large damages as ships built over sixty years—most say that the device was very appropriately named, from the inventor's standpoint.

All good businessmen and engineers seem to have limitless scientific curiosities, and are always anxious to try out the newest design-ideas and to redesign their own airplanes almost as soon as they are built. For a good salesman and capable manufacturer, Larry Bell seems to have a remarkable insight into the workings and possibilities of the minds of engineers. Fred Fisher, Asst. Chief Engineer of Curtiss, once told at a story about Larry Bell which illustrates the point.

A few years ago Larry was sitting in on an engineering meeting in which was being discussed the type of con-

struction to be used in a new design. The engineers couldn't agree at all on the type of wingplane to be used for the fuselage, each man advocating a different type. After a long discussion of pros and cons, Larry spoke up and said, "I know one type of fuselage I'm sure that all you engineers would agree to use!"

"What's that?" they asked.

"Any type that has never been tried before."

News features show Imperial Airways' reforms, awarded for trans-Atlantic service, being refused by page 11 in the six, from a flying gun tank—A.A.P. bomber being shown in. We suppose it won't be long before they've worked out some way of turning a group of machines into, to leave the warbirds, check the oil, and refill the radiators.

Our suggestion seems to have been a bit somewhat lately about the "game-maker" of the Pacific power and South America and as we are glad to know that there is some nothing to be swept about so far as July is concerned. The Dakota recently completed a "good will" flight down there,



and we know of nothing while gets more of a certain in airplane relations than a good will flight.

As A.P. insurance makes that a couple of British airplanes, out of which would be British insurance, in planning a flight to Japan from Rome. The route would be around the southeast side of Turkey, through Central Asia and over parts of China now under Japanese control, and the purpose of the flight would be to demonstrate the ability of flying from Italy to Japan without landing in territory controlled by Great Britain and France.

This business of flying cross country without receiving adequate military areas is becoming more difficult each day, and these areas and around boundaries are changing so fast lately we are wondering if it wouldn't be a good idea to broadcast information regarding them to pilots by radio along with the regular weather reports.

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ECONOMY THROUGH COOPERATION

It is no secret that the air lines are having hard sledding right now. All traffic routes are down, and competition on certain parts of the country is very keen for what little business there is during the winter months. Naturally, the traffic boys are scrambling for customers' wallets. The most disturbing element in the picture, however, is the swelling tide of drastic cuts and economies. Everywhere we hear of reduced schedules and layoff of personnel.

It is considered good business to retrench in bad times, but the real danger for the air lines is in cutting in the wrong places. No air transport operator can afford to cheapen the terms of maintenance and not to operating safety. No more short-sighted policy could possibly be adopted than to do anything that might open up the way for another accident at this time.

But there are others of economy open through cooperation if the air lines could only get together to work these out. Costly competitive traffic promotion campaigns could be tossed overboard in favor of a

joint publicity campaign that would put back traffic on all the air lines; less elaborate meals might be served in the air. Most important, there are available economies in joint action at many seasonal fields. Now that most operators are flying the same equipment, the way is open for a common spare parts pool at every point, thus reducing individual stock inventories. Already at some fields certain air lines are co-operating in the matter of common fire protection equipment. It might even be possible to make joint use of certain items of surviving equipment.

Some of these things are small in themselves, but careful study may open up real possibilities for savings. Who knows, with the correct trend toward standardizing of equipment, we may even contemplate the prospect of a common flying equipment pool, and even one or two large servicing and overhaul centers to take care of repair and maintenance for all lines. Utopian perhaps, but possibly something to think about.

MERRY-GO-ROUND AROUND

WARRIOR AIR LINES of the country are struggling to keep their heads above water in what is perhaps the most critical period of their existence (not excepting February, 1934), aviation affairs continue to get a terrible towing about in Washington. When several months ago we had begun to think that there was some chance of getting the several factions together on some definite plan for the establishment

of a broad gauge aviation policy for the United States, now things seem to be more at issue and more than ever before. It is a rather appalling thought that the whole future of this great industry (or even the future of the country itself, if you think of aviation in terms of national defense) is now being kicked about from congressional sentiment to congressional sentiment as a good old-fashioned political football.



Clearly there are too many grandstands in the game—too many people who want to carry the ball when they lose a disc field ahead and a chance to enter a pain for themselves, but who are more than willing to pass it to someone else when the going gets tough, or the crowd is not watching.

We had few hopes that anything important for aviation would get through the late featured special sessions, and we were not disappointed. We did look forward, however, to getting somewhere in the current regular session, but it looks now as though we are going to be disappointed again. It looks like the same old every-go-round-around again.

Meanwhile the domestic air line situation is critical, but line operators are jittery over the effects of the CAR on their business, foreign companies are threatening our position in world transport and world aircraft markets, and our service aviation is being neglected. If something isn't done soon to work out a coordinated national policy, things will certainly go badly in 1961. Responsibility lies squarely on the White House doorstep. The old game is at one of its turning points right now. Think fast, Mr. Quarterback!

THAT WEATHER AGAIN

BRANDER before a meeting of the Institute of the Aeronautical Sciences in New York, last month, Tony Tomlinson, TWA's high altitude researcher, had an idea down on the table that is worth looking at. After pointing out that the old theory that weather conditions became normally uniform in the sub-stratosphere was far from true, he suggested the need of making regular weather flights at a series of points around the country at altitudes up to 30,000 feet or better instead of the commonly accepted 15,000 or 20,000 feet. Obviously in order to do this, special equipment is required, but he suggested that there was a place where the Army's super-charged single seat pursuits could be used to advantage after they have become obsolete for military service. Not only would the government be able to recover considerable return on an investment that will have to be written off anyway, but also the Army pilots who would be assigned to fly regularly to such high

altitudes would receive a great deal of valuable experience in sub-stratosphere operations. Besides, the data that they would bring down with them daily would be of tremendous value to the boys who dip up the weather forecasts.

GIVE AC RESERVE A BREAK

CRUISE over the country, our attention has been called more than once to the difference between the operations of the Air Corps and the Navy's Reserve (as far as flight operations are concerned)—with the score very much in favor of the Navy. Not that the Army boys are less enthusiastic, or less competent individually, but, with the present set up, they are pretty much behind the 8-ball from the start.

It is apparent that Naval Reserve Units are really units, that they are trained and ready to meet any emergency on a very short notice. The personnel is paid for the time they put in on a scale that takes into account and includes the usual 50 per cent extra for flight duty. Then too, pilots can count on plenty of flying time, with 45 hours of scheduled airplane flying required during the year (including at least ten hours of primary, more hours of bombing, seven or eight hours of tactical flying, etc.). During drill periods, the reservists take over and operate in their assigned areas on a full emergency basis.

With the Army, conditions are different. Reservists are not required to fly any specific time during the year, nor do they get any compensation for turning out for drills. The only return available is in flying time, and, for most units, that is usually restricted by budgetary limitations. While there are exceptions, generally speaking the effort to recruit in Air Corps Reserve units is more often than not based on that of Naval Reserve units.

The real trouble, of course, is lack of funds. It might be a very useful idea for the boys who hold the purse strings in Army headquarters at Washington to consider seriously the differences that exist between the two services, then to set to work to rectify the Army Flying Reserve on some similar footing to the Navy's. Certainly the Air Corps needs the same break.



"Drop-down" delivery of a Boeing exhibit on Hyde Field.

"CAVALCADE OF FLIGHT"

Chicago's International Aircraft Show Attracts the Public, Sells Pioneers.

WHEN Chicago's International Aircraft Show wound up its last session at midnight, February 6, a preliminary count of the gate indicated that more than 200,000 people had visited the exposition during the preceding ten days. It was evident not only that the public had gotten its money's worth, but, judging from the remarks of exhibitors up and down the floor, there was a general feeling that the Show had been worth while from the industry's standpoint.

There were complaints, of course, but by and large a general feeling of satisfaction seemed to prevail. Not only had airplanes and accessory exhibitors developed a number of valuable leads and prospects, but initial sales of aircraft were reported by Aviaton, Beech, Bellanca, Furell, Guller, Howard, Lockheed, Piper, Pratt & Whitney, Raytheon, Ryan, Stearns, Waco, and others.

Interesting, too, was the fact that the manufacturers of planes in the \$4,000 to \$5,000 class began to report active inquiries from people who had bought planes in the \$1,000 to \$1,500 class in the preceding year or two. In other words, it is becoming appar-

ent that the light plane is actually beginning to develop business for the 4 to 6 place plane rather than on into their sales as was feared at first. Aviaton's "Impinger" Register listed in the following way statistics on aircraft in the Show:

Total number of planes on display, 45. Total number of commercial airplanes carrying ATC's as follows: high wing monoplane, single engine, 14; high wing monoplane, single engine, 5; single engine biplane, 4; amphibious single engine biplane, 1; single engine low wing monoplane, 2. Two cargo law wing monos, 4. In the non-ATC group, 11.

In the military group were 2 single engine biplanes, 2 single engine low wing monoplanes, and 1 motorboat. Largest ship in the show was a Douglas DST exhibited jointly by American Airlines, TWA, United Airlines, and Pan American. From early morning until late each night show visitors filed through the city, most of them first getting their first look at how this transport vehicle had really grown up. Surmounting the Douglas in the main arena was four

winches, three in which flamed (1) a Coast Guard Grumman amphibian, (2) a Bellanca DGA, (3) a Beech Bonanza, and (4) an Aviaton S, the latter three on Hyde field.

Heroes of Air Commerce and SACA exhibits attracted considerable attention. There were the "prop-down" type where the customers could push buttons and see what makes this aviation business tick. Also dynamic phenomena and Army jets were featured. Outstanding were the working models of the full-scale tunnel and the high-speed towing tank.

Most visitors seemed to enjoy John Ray's "The Airplane" program "Cavalade of Flight." In two performances daily were generally packed.

One feature that appealed to exhibitors was the fact that a landing strip was available alongside the Amphitheater on which most of the ships were tested (including the Coast Guard Grumman and the Lockheed) under their own power. In spite of turbulent weather and a somewhat disorganized approach and landing area, all flights were completed without accident.



SHOW SCENES



J. A. Sperry and Owen Wheeler of Aviation Publishers.

Sperry showed off his work on a potential passenger system.



Sperry showed off his work on a potential passenger system.

We couldn't blame Jim O'Brien for his enthusiasm for the Air Mail.



AVIATION
March, 1939

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Aviation featured in transparent panel and a set of instruments with transparent cases to show how the whole system works.



Jim O'Brien, editor of the *Aviation*, and the advertisement of Best Sperry.

Sperry's love was going to selling Air Mail for Jim O'Brien.



Sperry looked up on interesting working model of the automatic pilot with a related display of air line position.



At table of the Radio Company, Inc.



Dick Webb, Editor's What Coast representative (left) stands with Henry Howard.

The Radio booth displayed the company's full range of products.



Sperry showed off a few of the features of his proposed new system. Tuller, DGA is in the foreground with two special Air Mail (left) engine (right) and (right) Radio Company booth.



A customer "jazz de work" at the Radio Engine Company booth.





He Covers the WATERFRONT

**\$10,000 a year from a seaplane operation
in the heart of the Blue Ridge Mountains**

DONE AGAIN, you recall of West Virginia, where trout, musk, and an occasional bear are about the only things an airplane can find in East Tennessee's an operator with an Aeromarine 300 biplane who in five years has taken 129 students three seven thousand hours and earned a gross income from flying operations alone of between \$15,000.00 and \$20,000.00. Incredible as it may seem, his story can easily be checked and as easily appreciated by anyone who takes the trouble to journey to Charleston, W. Va., and ride the first man he sees about Glenn T. Clark and his seaplane service. As a matter of fact, it is quite obvious that this point about every one knowing him and commonly seeing his ship in the air over him, his last more than a 100 to do with his success.

Way back in 1932 Clark was flying at the best airport—some 175 miles to the 1920s. Things didn't seem to be passing on and he decided to see if the idea of seaplanes

openness, which he had read about, couldn't be made to work. Flying as captained he took up a flying ship program to finance the ship, and purchased an Aeromarine, one of the very first light planes to be licensed on floats. The Knoxville River provided him, line of ships, with a very nice airport, and he maintained an old air base and founded it as a charter base which had been abandoned for twenty years. The total cost was \$500.00 cash, plus a return amount of his own good common sense and labor. That same happy and happy is still in use today but he expects to sell, and replace it this Winter with larger quarters of the same general type but with the addition of a boat garden some painted glass and a compass on the roof. Clark has noticed that the poor people like to walk down to the air base to watch his ship fly, and he intends to make things more comfortable for them and incentive his income at the same time.

The Knoxville Seaplane Service,

"Blue" at Knoxville & Capital Service, is steady a block from the main U. S. O. highway bridge, and about 100 feet or so below the Union Bridge—the oldest in town. Locally, it is right at the center of things, and quite naturally a lot more conveniently reached from the airport, by automobile and on foot. The customers like it, moreover, because it is not only low all day to fly from water, but because good landing fields are scarce in that part of West Virginia.

As they go down there, you can find much to land on except trees, rocks and water, and passengers are low on all these—particularly the water!

Clark kept his original Aeromarine for three years and it logged some 3700 hours. During the third year he added a second seaplane but sold it after 700 hours because it was too expensive to operate and not particularly popular. He is flying his third Aeromarine now which he purchased

| Day | Altitude | Time | Temp. | Wind | Barometer | Remarks | Weather |
|-----|----------|-------|-------|------|-----------|---------|---------|
| 1 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 2 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 3 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 4 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 5 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 6 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 7 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 8 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 9 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 10 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 11 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 12 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 13 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 14 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 15 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 16 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 17 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 18 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 19 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 20 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 21 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 22 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 23 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 24 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 25 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 26 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 27 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 28 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 29 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |
| 30 | 1000 | 10:00 | 50.0 | 10.0 | 30.0 | | |

new last June, and had nearly broken it in by October with a log of nearly 600 revenue hours. Just in case lost a spot, a pair or two from his log book is reproduced—the 30 day period of 197 hours which commenced when he put his latest ship on floats. His price for instruction or student solo time is \$7.50 per hour, while he averages between \$10.00 and \$15.00 an hour on passenger flights.

Clark paid off the members of his flying club sometime ago and is running the service out as business alone. He doesn't advertise, doesn't line up the posters, and doesn't put literature on an easel in the hangar. As a matter of fact, his location and the fact that he is operating a seaplane seem to be responsible for bringing in all the customers which he has and has personally trained assistant can easily handle, but he is planning on a second ship with a definite expansion in all related activities next year, and at that time will probably go either business or more nearly the original manner. It's quite obvious, however, that every move he has made had a purpose. Aerial photography, the outboard, backed up by a fixed main engine and fixed propeller, mutually added to the income from flying operations. During the floods last Spring he carried doctors, nurses, and other important over the sea-

dated area, while for days at a time he and his ship were the sole means of communication with many towns along the river. In many cases, of emergency his immediately available service has obviously been an asset to his community.

As a side note, Clark runs a steady motor boat service. On many days he builds boats, and overhauled engines for the 5 or 10 customers who take up their work at his boat and store these motors and gear in a part of his hangar. It's just an additional feature which ties in with his flying and adds to the \$10,000.00 a year income from his steady seaplane activities. Another source of his income is the stationary mill which he owns, his equipment and handles his operations. The hangar, for instance, which is built on one end of the barge, is only part of his income to hold the ship if parked in a hangar, since it is limited by the width of the barge. For this reason the hangar is built on the side of the open end, and the ship backed out on its lands and upon a very low low slanted dolly, which serves it sideways through the hangar door.

There's a very low float, along the open side of the barge, made of 12"x12" split lumber. It supports the lower part of the ramp



Glenn Clark knows that airplanes will never take when they are in the air. He has been in the air for 10 years and has a 10 day pilot.

which is hinged at the dock, and makes a good thing for students to enter without any confusion to improve, because a head on contact simply runs the job out of water—the less head on is right. Of course, it would be out as rough water but it works perfectly on the Knoxville. Incidentally, the barge is a wonderful rig to work from, is water, when inside and during changes in level take place, for Clark simply moves it up or down the levee for floods or droughts. Just that of getting away with a lot of time in a place that has forty foot tides, and always having things right.

There are literally hundreds of Charleston, West Virginia, around the country. Many of them, of course, after different operating conditions. In some parts it is possible that a larger, more powerful ship would be a better job, if usually enough water provides. In others it might be more economical to store the plane out of town, and fly it in every day to an operating base in the heart of the city. These are questions for the Glenn T. Clark to decide. But a quick glance at any map will convince any prospective operator that there is an almost unlimited amount of virgin territory where the West Virginia plan can be adapted. And in these days of low competition, with the writer attending the operators and private pilots call more and more airports, it is obvious that the seaplane idea is proving to be a good one just what a pilot will probably follow next year.



CHEYENNE

GLIMPSES

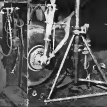
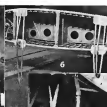
A picture trip through United Air Lines' efficient and modern overhaul base.

UNITED'S LONG ESTABLISHED MAINTENANCE CENTER AT Cheyenne (Wyo.) has not only had its face lifted by the rearranging of the exterior of its buildings, but within the past several months all shops have been rearranged and re-equipped, so that today it stands as an example of the best and most efficient in the servicing and overhaul of transport planes and engines. Presently every bit of equipment, large and small, at UAL's big hangar system passes through it, coming out again into service "as good as new."

The service hangar stands in approximate relation to the other shops as shown on these pages. It serves not only for routine inspections and day-to-day checks, but also as a disassembly and re-assembly department for overhauls. Thus the shops are stripped of engines, propellers, instruments, radio, battery, etc., and the parts distributed to the appropriate overhaul shops. A new Baker-Bulfinch crane type electric truck has been found very useful in handling such parts across to other buildings.

All overhaul shops have been laid out with an eye to straight-line production methods. The engine shop is especially noteworthy in this respect, with motor moving through from tear-down, cleaning, inspection, re-assembly into bus home, from crew to crew, efficiently and smoothly aided by a modern overhead trolley system.

An interesting project is being completed in both the instrument and radio shops. All test equipment is being standardized and mounted permanently on standard safety panels, so that should they wear out of the world and wander into test setups that are new to be in in any overhaul shop.



1. Shop business here has been streamlined to maintenance as well as to manufacturing.

2. The welding shop is well lighted. Shovele frame layout kept clean at all times.

10-11-12. The propeller shop is well lighted, is well laid out with overhead rollers and portable crane for easy handling of large blades and hubs. Adjacent grinding and polishing rooms are well lighted by overhead indirect type in handle dust. The portable hydraulic hoist (12) has proven a most useful propeller handling gadget.

13. A portable blower with flexible hose can blow both the late arriving shops in hot days.

1. Cleanup of an automatic lathe for tapping plates done in minutes. Computered air function power for operation.

4. Each instrument man has own bench fitted with special tools, vacuum, air and electric units.

5. A gallery shop is well provided with floor-to-ceiling stainless-steel service benches.

6. Sublimation stands of steel tube keep wing panels off the floor and in working height.

7. A small motor operated vacuum pump deflates big tires for easy removal of tube and skin.

8. A divided electric motor provides power for both the two machines working.

9. Everything ahead of the fire walls may be assembled in place on these engine mount jigs before arriving under wings into service.





SOLVING SPACE STRUCTURES GRAPHICALLY

Part II Application of the Method

By Elias Moness

Engineer, Investment Research Department
Douglas Aircraft Company

A SPACE STRUCTURE members will be solved graphically by means of the method explained before. (See *ARTICLE*, December, 1947, page 56.)

The four trusses making up the structure are all different from one another. (See Fig. 4 and 5.) Therefore, we must show separate views for each truss, the side trusses are shown in the same view with the members of the other trusses shown in dotted lines whenever they do not coincide with the corresponding lines of the other side truss. The bottom truss is shown in plan view, that is it is easier to keep the direction of applied loads correct.

It will be seen that some of the trusses are plane structures. This makes it difficult if not impossible to use the ordinary graphical method in which each separate truss is dealt with as a plane structure.

The present methods differ basically from the above by treating the structure as a true space structure and by

dealing with all the members at any point simultaneously, regardless of their direction or of the planes they are in.

Spine loads may be copied at any point by means of the two projections which are necessary to represent them completely.

It must be emphasized that in space cases any one view will not show all the members at a given point, it must be kept in mind that there are other members ending at this point and these are shown in some other views, as in and in realizing the space nature of each point it is preferable to draw a first-hand isometric sketch of the structure and to refer to this sketch in dealing with that particular point. (See Fig. 3.)

In the present case a space load is applied at each of the four end joints. The vertical loads at joints (1) and (2) are 1200 lb. acting down, those at (3) and (4) act in the opposite direction, the side loads at (1) and (7)

are 1200 lb. and are outwardly those at (3) and (4) act inward.

The loads at the four joints are laid off each and their direction is marked by arrow heads.

The following three operations are performed over and over again and are therefore explained in detail.

1 In finding the stresses in two members at a joint for a given resultant R , the method employed is to draw a line parallel to one member then the tail end of R and another parallel to the other member, then the head end of R . Joint 1 shows two possible ways of doing this.

2 A single rule for determining the nature of the stress (tension or compression) is as follows: (See Fig. 2.) When the line of action of the load falls between the two members (or their prolongations) the stress in both is of the same character, (tension in case A and compression in case B). When the line of action coincides with one of the members, that member carries the entire load, while the line in the other member is zero, (Case A and B). When the line of action is outside the members (as these principles together) one member will be in tension and the other in compression (case C and F). Inspection will show which stress is in each member.

WHEN applying the various components of the first load acting at a given joint, care must be taken that the tail end of any component coincide with the joint and at the previously applied component, as shown in Fig. 3.

The resultant R is drawn from the joint to the front end of the last load vector and points toward the latter. We can now proceed to solve the structure.

Starting the solution at (1) we notice that this joint has three members A , C and B , in the side view A and C are perpendicular to each other while B is the angle member. This joint can be solved in two ways: we draw then the front and at the head corner a line parallel to A and C and obtain directly the side projection of the stress in C , this intersection point A , is projected in the top view and marked with a small circle. The stress in C is compressive, arrows are placed at each end of C pointing towards the respective joints. At this moment it is noted to remember that the stress in C will appear as an applied load at joint (10). It is therefore advisable to make off with the end of the stress at joint (11). A little circle will mark its magnitude, arrows at each end pointing towards (11) will indicate that the length of the stress vector is to be measured between the circle and the joint (11).

The horizontal projection of R is applied in a phantom load at (1) in the top view. It is drawn to the left, as a stress line, since the phantom member A and B in the side view was in tension. A small circle marks the end point of this applied phantom load in answer for the load F , as drawn then the last point a line parallel to F , and by off this load on the line. Another small circle marks the end point of the load. To answer for the side component of the stress in C we draw a horizontal line (1) in the top view and project on it the stress in C in the top view. It is for this that we marked the point A , when we found C in the side view. The side component is A , and it acts outward (since C is in compression, and the stress in the top view would point towards the support side).

The resultant of the three applied components is found by drawing a line from the joint to the last circle drawn. (The end of the last component.)

An arrow is used to mark the direction of this resultant, and a check mark is made on C to make it easily recognizable among the other lines at this joint. A line parallel to R is drawn then the front end of R is projected to its intersection with A , as before, the length A , is marked off and laid off at joint (6). A small circle with the two small arcs referred to before mark its magnitude, arrows at each end of A and B show compression in A and tension in B . It is laying off R at the joint (1) it will be noted that the stress vector is longer than the member. The member B is, there-

fore, extended (by a right line) beyond joint (1) and the small circle is marked on this extension line; a circle with arcs as an extension line always indicates that the stress is to be measured to the farther end of the member.

It is preferable not to make the stresses right away, but to measure the graphical solution in its very end and then to proceed with the making of all the stresses.

Joint (2) is solved very much like joint (1) but the direction of the vertical applied load being opposite to that of F , the sign of the stresses in C , A and B are all opposite to those in the corresponding member at (1). The side load has the same direction in both joints, but the stress here is inward at (2). Our solution at joint (3) and (4).

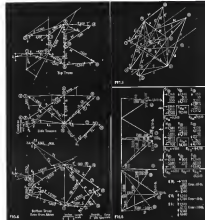
THE REMAINERS of the upper truss is the first by a plane structure, and it is convenient to continue right now with its solution. Joint (8) has three members, we cannot solve it unless we first solve joint (7) which has only two members. There is only one applied load at (7) and that is the compression in B , a line drawn

from the front end of this stress vector and parallel to member F and another line from the tail of the stress and parallel to IT give the respective stresses. The IT stress vector is longer than the member, both members are in compression. Now that F is an applied load at (5) we have only two members at (4) and we can solve that joint. Fig. 4 shows how this is done.

Note the manner in which the resultant applied load is found, how its direction is marked, the distinguishing check mark and the solution for the stresses in G and H .

We cannot proceed to solve joint (5) or (4) because each of them has more than three members (see caution in sketch). We therefore return to the front end of the structure and solve joints (5) and (6).

Joint (5) is very much like joint (1), it is side view in the side member, F is the side projection of the stress in C , as the bottom view the resultant contains the horizontal projection of F , the load F and the side component of the stress in C , a line drawn A parallel to F gives the stresses in F and B . These are marked and laid off at joints (5) and (10) respectively. Joint (4) is (Then to page 60)



Putting the landing traction on the upper section of the wing before removal from the jig.

DC-4 Takes Shape

Rapidly nearing completion at the Douglas plant at Santa Monica is the first of the 4-engined land transports engineered by and sponsored for jointly by United, TWA, American, North American and Pan American.

A PROGRESS IS BEING MADE TO PUT Harry Wetzel or Arthur Raymond or any other Douglas official down to anything more definite than "early this spring" for flight tests on the DC-4, it is a known fact (and we offer these pictures as evidence) that the construction work on the big ship is better than 90 per cent complete. It is a fair guess that the big job may be out of doors and into the air some time during April.

The vital statistics on the ship (span 136 ft. 3 in., length 107 ft., height 24 ft.) are impressive enough, but a glance at the photograph of a couple of scale models on the opposite page tells the story more graphically. Its gross weight is 54,000 lb., or nearly three times that of the pre-

ced DC-3. Useful load is expected to be 20,000 lb., which will take care of 42 passengers, a crew of five, 5,500 lb. of mail, express and baggage plus fuel. Although no figure has been given as yet, its top speed will very likely be well above the 300 m.p.h. mark as critical design. Design landing speed is 165 m.p.h.

The prototype machine will make its initial flights and will undergo its

prototype tests with four Pratt & Whitney twin-turbine engines, developing a total of 3,500 hp. for take-off. Provision has been made in the engine unit in the nacelle for alternate installation of Wright engines.

Although the ship will be tested as a normal airplane, provision has been made for supercharging the engines of production models to maintain low level cruising pressures at high altitudes.

Some 12 "believe it or not" oil dem. Cost of development to date of completion of prototype, \$1,500,000.

More than 300,000 engineering man-hours are tied up in the first machine.

Available take-off power exceeds by approximately 1,000 horses the output of most of America's finest streamlined locomotives.

With one engine dead, the DC-4 can maintain level flight and clear the highest mountain in the United States by nearly 5,000 ft.

The Douglas testing laboratories were busy for more than two years developing required information on materials and design, more than 300 major structural tests in destruction were conducted, over 27,000 engineering and shop hours were used in plans and to check engineering designs and stress calculations.

Support for the DC-4 scale model up to a single plane.



By way of contrast, scale models of the DC-3 and DC-6. Note difference in attitude on the ground due to the elevator landing gear.

General view of the Douglas shop showing fuselage and wings in the assembly line.



Looking forward along the bulk as it lies in the assembly jig.

One of the 16 in. diameter main landing wheels fits into this recess in the under part of the outer section.



Note typical Douglas wheel shroud design, also the space under the shroud for the main wheel mounting.



Federal Trainer

Two-Place Military Primary Trainer Built to Air Corps Specifications now available for export

A REPRESENTATIVE FEDERAL TRAINING plane with a number of outstanding structural characteristics has been designed and built by the Federal Aircraft Corp. of Lincoln, Iowa, U. S. A. under the supervision of L. Charles Cox and Maxey McGraw. The machine is powered with a Wright 700 KT Whiskard engine developing 225 hp. at 1500 r.p.m.

The fuselage is in two parts bolted

together just aft of the rear cockpit. The forward portion is of welded steel tube covered entirely by sheet aluminum alloy. Landing features in the fuselage are attached by means of Dural fasteners, easily removable to give access to controls or equipment for maintenance and adjustment. The rear portion of the fuselage is of a semi-monocoque, all-aluminum alloy construction.

The landing gear is full cantilever type hinged at the lower fuselage. A torque arm extends vertically upward from the upper end of each leg and chain are attached to the torque arm and to the landing struts at the outer end of the struts just below the thrust line. The legs are fitted with quickly detachable clevis-type shock absorbers. General Structures fitted with mechanical linkages are fitted. The tail-wheel is also of General with standard air type shock absorber.

The wings are constructed of 24 ST aluminum sheet, fully covered. The leading edges are covered with aluminum sheet, riveted in the front spar and to the inner ribs. Flexible drag wire bracing is installed. Afters are in the lower wing panels only and

are inattachable. Interesting construction features is the retractable wing tips which are constructed separately and are attached to the main panels by external tension bolts. Upper and lower wing tips are retractable in case of damage to the wing tips. They are easily removed and replaced. All hinge pins and shackle bolts for main assemblies are either inside the skin or just inside the skin line so that no cover plates need be removed or access doors opened for inspection and maintenance.

The wing bracing is composed of a single 1/2" steel wire X type later plane struts. This arrangement makes rigging much more positive, even though the use of wire.

In test flights at Wright Field (Dayton) it was reported that the machine was very maneuverable and although it will not spin voluntarily, it may be spun deliberately with ease and positive recovery. It is said to be able to clear a 30-ft. obstacle after a run of 750 ft.

The specifications and performance figures given by the manufacturer for a gross weight of 3275 lb. (including two pilots and 40 gal of fuel) are as follows:

Wright engine 225 hp.
Fuel tank 40 gal.

Best cruise 180 mph
Cruise 180 mph
Upper wing span 33 ft.
Lower wing span 30 ft.
Total wing area 1100 sq ft.
Overall length 27 ft. 3 in.
Overall height 10 ft. 3 in.
Ground clearance (main) 120 in.

Maximum speed (sea level) 120 mph
Stalling speed (sea level) 40 mph
Rate of climb (sea level) 400 ft per min.
Service ceiling 10000 ft.
Endurance at operating speed 4.00 hr.
Range at operating speed 400 mi.
Maximum range 500 mi.



Noorduyn Norseman

Well known American designer turns out a land wonder for Canadian service that is attracting wide attention among bush-woods operators.



DESIGNED AND BUILT to meet the wide variety of conditions encountered by the commercial operator in Northern Canada, the Noorduyn Norseman is now offered in an improved version featuring many minor refinements incorporated as the result of operating experience.

The "Norseman" made by Noorduyn Aircraft Ltd., Montreal, Canada, is primarily a single engine, high-wing, strut-braced cabin monoplane all metal construction. Normal passenger capacity is nine plus the pilot. Although the Pratt & Whitney Wasp Type S1M1 of 150 hp is standard equipment, the Wright R1820 engine of 425 hp is also offered, together with the Hamilton-Standard convertible propeller.

Not only is the "Norseman" available with two different power plants, but it is also designed for use from

land, snow, or water, being optimally fitted with wheels, skis, or floats. Also in line with its all-purpose nature the "Norseman" has been built to operate as either a freight or passenger plane, with further provision for service as an ambulance plane, for aerial photography, troop transport, or for home private owner accommodations.

Specifications in a nutshell:
Span—54 ft. 6 in.
Length—32 ft.
Height—15 ft. 3 in.
Wing area—335 sq. ft.
Max. lift capacity—3075 lbs. as freighter, 2140 lbs. as passenger plane.
Disposable load—2175 lbs. as freighter, 2117 lbs. as passenger plane.
Gross weight—5440 lbs.
Powerplant—Pratt & Whitney S1M1 engine—150 h.p. or 425 h.p.
High speed at 1000 ft.—190 mph.
Cruise speed at 1000 ft.—138 mph.
Landing speed, full load—55 mph.
Range (usable with fuel)—700 to 1400 miles—normal full load range, 500 miles.

AVIATION
March, 1932

Now... all industry can make new production profits with the screw that

Gains Time

Guides Driver

Guards Work



Faster work with half the pressure

The best starting and driving — often best first try — with "Phillips" gives the operator of a standard screw driver driving metal one time less a trouble.

As easy as pointing your finger

The obvious guide for the screw's recessed head is the driver's guide — the Phillips screw driver's guide — the Phillips screw driver's guide — the Phillips screw driver's guide.

Better work — and no spoilage

The driver's guide also acts as the screw's guide — the Phillips screw driver's guide — the Phillips screw driver's guide — the Phillips screw driver's guide.

Given the test by certain key firms in the screw-using industries... shows where production costs are slashed with an eagle eye — where quality is right and there's no excuse for waste.

The Phillips Screw with the patented recessed head has been designed, after all sorts of tests, motion and quality studies, by engineers, engineers, purchasing managers, workers.

Now the production facilities of the screw manufacturers have been ordered — and we're in a position to meet the demands of any plant that has use for screws.

Take the job the picture shows. There the Phillips Screw gives time — and screws — and strength — and spoilage — and money.

Faster work. Use power drivers or higher speeds where power is already used. No need to study the screw, it

stands itself. No horns to remove afterwards.

Grosser holding power — assemblies are more solid, and often lower or smaller diameter screws will do as well. No broken heads — no screws dropped. Grosser driver and bit life — driver time saved.

Better work. The screw helps out. No push and turn — just turn. Maximum contact — maximum strength. Men are freer toward the end of day.

Better looks. The driver can't hang from a Phillips Screw to dig a channel across the work. No grooved slots — no cornered fingers. And the screw is flush with the surface.

What's your strongest problem? Find out what the screw with the patented recessed head can do. Send the coupon to one of the firms listed below... mention the type of work you expect screws to do... they'll send you folder A, telling you how well the Phillips Screw with the patented recessed head has done that work for others.

This is the Patented Phillips Recessed Head

Now the tapered slot — its angle was worked out after months of work to have low to reduce the driver's maximum turning power. With the flat bottom — its angle except in the slot. This prevents the driver from bending the screw. Manufactured to tolerance of plus or minus .007" — it guarantees a perfect fit. Head, eye, fit, or power drivers may be used — available from leading manufacturers and distributors.



MACHINE SCREWS

SHEET METAL SCREWS

STOVE BOLTS



WOOD SCREWS

PHILLIPS SCREWS

Guide Time... Guide Driver... Guard Work

PHILADELPHIA, PITTSBURGH, CLEVELAND, CINCINNATI, COLUMBIA, DAYTON, DETROIT, INDIANAPOLIS, KANSAS CITY, LANSING, LOS ANGELES, MINNEAPOLIS, NEW YORK, PHILADELPHIA, RICHMOND, ST. LOUIS, ST. PAUL, TAMPA, WASHINGTON, D.C.

American Screw Company, Limited, Philadelphia, Pa.

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Continental Screw Co.
New Bedford, Mass.

Garble Screw Corp.
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National Screw & Mfg. Co.
Cleveland, Ohio

Harwin-Kelton Corp.
New York, New York

SEND THIS COUPON TO ONE OF THE FIRMS LISTED AT RIGHT — FOR SPECIFIC FACTS ON PHILLIPS SCREWS — NO OBLIGATION

Name _____ Firm _____
Company _____
Address _____ Product _____





● In ships like the one pictured here there is upwards of 100 miles of wire in the control cables. If it is Hazard Aircraft Cable (either tinned or galvanized, or the 18/8 stainless steel "KOBOLDLESS")—every foot of that wire is drawn with meticulous care. We hold our diameter tolerance, for instance, to 5/10,000th of an inch! Tensile strength requirements are equally rigid. Small wonder aircraft everywhere are more and more depending on the uniform, consistent dependability of Hazard cables.

Let us tell you about the practically non-magnetic qualities of Hazard "KOBOLDLESS." Let us explain how the pyroforming process makes a far superior cable or strand. Write for full details—today.

HAZARD WIRE ROPE DIVISION

ESTABLISHED 1906

AMERICAN CHAIN & CABLE COMPANY, INC.
WHEELING, WEST VIRGINIA

General Offices: Pittsburgh, Chicago, Birmingham, Denver, Los Angeles,
San Francisco, Philadelphia, Fort Worth, Toronto, New York



BUY ACCO QUALITY

A FEW OF THE 137 AMERICAN CHAIN & CABLE INDUSTRIAL PRODUCTS

AMERICAN CHAIN DIVISION
DOMESTIC CHAIN (STEEL), 3/16 IN. TO 12 IN.
Welded Chain • Whitlock and Washburn
Chain • Ball-Bearing Chain
Chain-Machine Subdivisions

ARMSTRONG & CAMPBELL DIVISION
All sizes Galvanized and Stainless
Special machinery • Milling machines
PUMP-CHAIN BLOCK DIVISION
Chain Hooks • Ropes

HAZARD WIRE ROPE DIVISION
See full literature "Wire Rope & Ballistics"
Wire Rope • Improved Spring Wire Wire
Rope • Steel Ball Cable

HIGH-LAND ROPE & WIRE DIVISION
Wrought-Iron Rope and Steels

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Fence Posts • Wire and Ball Products
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Pulley • Electric Steel Springs

HEATING WIRE, CASTING DIVISION
Electric Steel Castings, Rough or Finished
Industrial Steel Castings

WIRE NET MANUFACTURING DIVISION
Chain Netting • Electric Welded Chain

An Assurance for Great Safety

HAZARD Aircraft CABLE

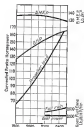
ADVERTISING
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A NEW Approved Type Certificate (187) has been awarded the Ranger Engineering Corporation, rating the current production diesel engine at 365 horsepower. This new model, designated as the 6-410B engine, is the latest development of the original Ranger 6-380 model.

Now, as a major event, parts on all Ranger engines are interchangeable. It has been possible to incorporate many of the refinements which were part of the high-power twelve-cylinder development in the latest six-cylinder engine. In addition, the cylinder bore of the new engine has been increased from 4 in. to 4 1/2 in., increasing the displacement from 385.4 to 412 cu. in.

The induction system has been modified by the use of 30 deg. intake valves and the new Stromberg NA-18-0 carburetor. The compression ratio was raised from 6 to 4 1/2:1 and the engine rated on 80 octane fuel. Improvements have been made in the valve gear, the most important of which has been the change in the type of valve-rocker adjusting screw.



The mild spherical-end screw has been replaced by a one-piece screw with a ball-end ball-end and a spherical end. Turn and service experience have proved that this new design will eliminate valve tip jacking and greatly reduce wear at the point and in the rocker shaft bearings. Valve tappets will maintain proper clearance over long periods of operation without any inspection or adjustment.

The entire engine, including all parts of the valve operating mechanism, has positive automatic lubrication without the use of any external source of pipes or connections. The lubrication system is equipped with effective built-in oil-throwing centrifugal oil-collecting devices. No parts are so constructed, except pressure-reducing ball bearings, all operating parts being fully enclosed, and all accessories being of the self-lubricated type.

The 6-410B is equipped with two flexible, 5500 R.P.M. centrifugal magnets. This unit has piston-type contact lockers and sealed ground-proof ball bearings. The magnets operate on lubrication between major overhauls.

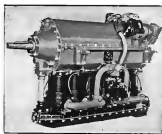
In the small engine, powered by an engine which may be operated full throttle at sea level, the take-off performance of the airplane is largely

affected by the form of the Ductless Exhaust Pressure curve for the engine. This is particularly true in multistage wing load-grip propulsion.

To obtain the best possible take-off performance, it is desirable to have the take-off engine, coupled with the engine, the maximum R.M.P.P. The actual drop in rpm from rated to take-off conditions with any load propeller is governed by the difference in B.M.E.P. between rated and take-off rpm's. The larger this difference, the less the drop in engine speed during take-off.

One means of obtaining the optimum take-off power is by having the valve overlap relatively small, giving improved volumetric efficiency at the lower r.p.m.'s. This condition has been achieved in the new Ranger 6-410B engine. With 55 deg. overlap, the B.M.E.P. has been brought up to 150 lb. per sq. in. at take-off r.p.m., and the horsepower at take-off raised to 365 hp with a rated maximum of 385 hp. The result is a marked improvement in both take-off and climb performance over that obtained with earlier models, which provided a take-off B.M.E.P. of only 125 lb. per sq. in.

The accuracy shown on the 6-410B



RANGER 6-410-B

A.T.C. based on new 365 hp. in 28 sec.

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41

(Continued from page 41)

is the same type as used on the high-speed models. The down air flows from front end of the cowling at a point of minimum torsional vibration and transmitted through a single bellows shaft to actuate three gears at the rear, the inherent flexibility of this shaft eliminating the transmission of vibrations and shock loads to the structure in their driving gears. The engine is equipped with steering and down air control paths for vector generator, fuel pump, and tachometer.

A.T.C. No. 116
Base (14,000 ft.) (maximum) 116
Stall (14,000 ft.) (maximum) 84 in
Climb rate (14,000 ft.) 412 ft. in
Cruise rate (14,000 ft.) 635 ft. in
Fuel (14,000 ft.) 80 Gallons
Reduction in level performance
Full Throttle
Stroke horsepower (maximum) 180
r.p.m. (14,000 ft.) 2450
Recommended Cooling Limits
Stroke horsepower (maximum) 118
r.p.m. (14,000 ft.) 2250
Dry Weight (with standard equipment) 250 lb.



STINSON SR9-A

A new Reliant invades the medium priced field

Power at \$18500 at Wright, Michigan, to compete with similar two and three passenger planes, the new passenger Model SR9-A Stinson was shown publicly for the first time at the Chicago Air Show. The Model A is a full sized Reliant powered with the 225 HP Lycoming Motor, which drives a Hartzel metal air-screw propeller. The motor is equipped with automatic valve heli-copter, which makes the engine to drive 14,000 miles between overhaul. Standard equipment of the Model A Reliant includes in three features commonly associated with Reliants in the past and includes electric starting, hydraulic landing, folding wing vanes flap; rounded safety glass windshield and other windows also of safety glass; wheel, full overdrive landing gear with spring brake shock absorbers, all metal internal construction in wings and fuselage; tail NACA Cowling wheel pants and full flaring throughout.

Extensive finish is in a customary Stinson standard. The fuselage is dark blue, highly polished, with a one leading wheel stripe. The pulled wings are trimmed in a design dominated by silver with a dark blue contrast.

The sales is designed to give plenty of room for the passenger. In fact, the rear seat is the same width found in many cars where those persons are carried in the rear seat. Back seat pilot seat can be adjusted five and all and the seat backs may be set at different reclining angles. Upholstery is cloth throughout. Controlled ventilation and cabin heater, aerial search light and sub wings are provided. Dual controls are standard equipment, as are navigation lights, tail and landing indicators, instruments for mounted as a shock proof unit, fully hydraulic dual and include Kollsman air speed, compass, altimeter, barometer, oil pressure and suspension groups.

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Franklin AC-130

(Continued from page 34)

is designed as an integral part of the shaft.

Another innovation of this Franklin AC-130 engine is the use of "Dilettante" timing gears. It was found that these gears are definitely more quiet and longer-lived.

The oil capacity of the crankcase is four quarts and the maximum consumption of the engine is approximately one ounce of oil per hour when operating at 2400 r.p.m. under full throttle operation.

The alloy steel connecting rods are of international design with a bronze bearing placed on the upper end for the piston and lubricated by splash within the engine case. The big end bearing is of the replaceable pin rod type with a copper lead lining.

Pistons are Ray Day designed and cast of aluminum alloy. The piston pins are full floating and are secured by an alloy steel spring at each end. The piston carries Perfect Guide 3 rods, Type 200 rings in the top groove, a Single 1 inch standard ring in the next groove. The pistons are fitted with 5025 inch clearance.

Carburetor is by a Stromberg MA-52 up-drift carburetor with 1/16 inch venturi and No. 32 main jet. Aluminum intake pipes connect the manifold separately to all four intake ports.

The Franklin AC-130 engine is regularly supplied with single Scotch magnets with single plug. In its two cylinder with dual magnet, or with single Scotch and Aviatone pistons, or with dual Aviatone pistons.

A self-contained cooling system can also be supplied for this engine—or it can be cooled from the propeller. This self-contained system incorporates a centrifugal pump mounted on the crankshaft and suitable hoses for circulating air to each cylinder in correct proportion.

General Specifications

Price \$14,000
Displacement 180 cu. in.
Compression ratio 10:1
Fuel Gasoline 70 octane, 25 octane or better recommended.
Power 180 hp at 2400 r.p.m.
Weight 150 lb., engine option.
With service 240
Length overall 27"
Height 14"
Width 14"
Remarks Four papers issued with the crankcase, 10 lb. alloy, 100 in long shaft.



ROYAL DUTCH AIR LINES

Super-Speed LOCKHEED 14 Powered by WRIGHT CYCLONES

Super-speed Lockheed 14 transports, powered by Wright Cyclones, are now being delivered to K.L.M. (Royal Dutch Air Lines) for its European and West Indies-South American operations—and to K.N.L.M. (Royal Netherlands Indian Airways) for use on the Dutch East Indies.

The new Cyclone-powered Lockheed 14 has a high speed of 260 m.p.h. These swift airplanes will provide Europe's fastest air transportation. They will be used on K.L.M. routes which speed in all directions from Amsterdam across London, Paris, Berlin, Vienna and many other important cities of Europe.

K.L.M. will also use Lockheed 14 transports on its major routes between the West Indies and South America—including from the island of Barbados to Trinidad and on to La Guayra, Caracas, Azua, and Manzanillo, Venezuela.

K.L.M. (Royal Dutch Air Lines) and its associate company K.N.L.M. (Royal Netherlands Indian Airways) is the world's greatest airline system—extending from Amsterdam, Holland to Batavia, Java—a distance of 9000 miles.

Wright Cyclones power all of K.L.M.'s great fleets of Douglas DC-2 and DC-3 airplanes. These giant planes fly from Amsterdam, Holland to Batavia, Java at 35 days—compared with 1 week by the fastest rail-water route. K.L.M. recently completed, with a Cyclone-powered Douglas DC-5, its 50th round trip to the Far East.



The fleet of the new Cyclone-powered Lockheed 14's determines K.L.M. (Royal Dutch Air Lines)



WRIGHT
AERONAUTICAL CORPORATION
PATENTED NEW ZEALAND
A DIVISION OF CHRISTIANSEN CORPORATION



Aircraft Radio

New Equipment for Communication and Navigation by Navy and Fleet



United "hoop-loop"

New device tested on ground stations

Experimentation by the recent of their state-of-the-art program in navigation with their planes, United Airlines have developed an electronically-shielded loop of large dimensions, to be used in ground stations. The loop, about 5 feet in diameter and of thin section, closely resembles a large hoop. The metallic tube, which surrounds the loop wire, prevents the electrostatic component of the incoming wave from affecting the receiver. The magnetic component, on the other hand, is reserved due to the presence of an induction segment in the tube. Since static is primarily an electrostatic effect, the loop discriminates against static in favor of the signal. The theory of this principle is slight in the special type of "hoop-loop" tube employed in the upper air, depends upon the strength of the static. Its effectiveness against ground static (noise static is rarely encountered on ground stations) is now the subject of study by United engineers.



BOB ARDINO & BOB Ruppberg now being tested by United on radio stations for ground stations.

Radio Navimeter

Navigation calculator for radio direction-finders

THE L.M.T. LABORATORIES OF PHOENIX, a subsidiary of the International Telephone and Telegraph Company, have designed a compact calculating instrument, called "Radio-Navimeter", which can be used for transmitting the estimated bearings of any navigation instrument (magnetic compass, gyro compass, or radio direction-finder) into the bearings relation to true geographical north or magnetic north. The device contains a cone mechanism which automatically tabulates the required corrections for the quadrant error of the radio-compass.

One side of the calculator contains data, separately-rotatable compass cards, use for geographical bearings, one for magnetic bearings, and one

for radio-compass bearings, the latter being compensated. Two sets, rotatable about the center, are used for lining up the three compass cards according to the requirements of the problem. On the inner side of the calculator is a small circular slide-rule, and tables of regular values needed for computing ground speeds and distances traveled. The instrument measures about 6 inches in diameter, and weighs about 10 ounces. The quadrant error can be set according to the requirements of the navigation work which the calculator is to be used. If the error curve is known, the pilot may set the same himself.

D of C Radio Report

Jackies Reviews
Autonomous Landing

ALL FLIGHT AND ASSOCIATED ENGINEERS presented in the autonomous-landing problem should share a copy of "Report on the Status of Instrument Landing Systems", written by W. E. Jackson, Chief of the Radio Development Section of the Bureau of Air Commerce, published by the Department of Commerce. The history and present state of the art are thoroughly and fairly treated, from the earliest Bureau of Standards work in 1919, to the 1937 recommendations of



New MARTIN 166 BOMBER

Now—the Greatest in a long line of famous military aircraft—the new Model 166 Martin Bomber . . . It is a Great Ship: Greater Speed, Greater Range, Greater Staking Power, the newest Martin Bomber has back of it the heritage of twenty years of experience in the development of bombardment aircraft for national defense the world over. Aerodynamically improvements and structural refinements give the Model 166 striking power undreamed of but a few short years ago . . . This new ship is available for export. We will be glad to quote delivery dates to interested governments.

THE GLENN L. MARTIN COMPANY
BALTIMORE, MARYLAND, U.S.A.

Builder of Dependability  Since 1910



KOLLSMAN

ELECTRICALLY HEATED PITOT-STATIC TUBE

ICK will not form on the critical surfaces of the Kollsman Electrically Heated Pitot-Static Tube—ensuring continuous accurate operation of the Air Speed Indicator, Altimeter and Vertical Speed Indicator under all flight conditions.

The design of the Kollsman Pitot-Static Tube—mechanical, aerodynamic and thermal—is based upon exhaustive wind-tunnel and flight tests. All elements are enclosed in a one-piece copper shell of correct aerodynamic form. The heating element is hermetically sealed within the shell. The tube itself is rain-proof and self-draining, both on the ground and in flight, and no amount of water can get it out of action.

Electric supply is automatically regulated, giving maximum heat at low temperatures, yet so reducing the current at high temperatures that it is impossible to burn out the heater. The electrical connections are permanently welded.

The Kollsman Electrically Heated Pitot-Static Tube, Type 172B, is for mounting, as shown, at the lower end of a vertical streamline tube. Two other types are available—Type 303, for the forward end of a horizontal round tube, and Type 399, for the upper end of a vertical streamline tube above wing or fuselage.

Specifications and more detailed information will gladly be supplied on request.

the aircraft. The report is 11 pages long, and though it is not illustrated, the author manages to make clear the methods used by the Bureau, by the Army, by Germany, by Russia, United Airlines, TWA, and Washington Institute of Technology. An excellent list of references closes the report. It is highly recommended.



AR-36-W Model AR-36-W Transmitter

Two New Transmitters

The "Shrove-Aero" and the AR-36-W for private flyer

ALL RADIO, of the Chicago Municipal Airport, at 5214 W. 63rd Street, Chicago, announces a new transmitter, Model AR-36-W, intended for private aircraft. The entire transmitter and power dynamo are mounted on one chassis, eliminating the need of extra power cables. Fixed, locked tuning is employed, crystal controlled, with the standard 350 kc. channel frequency. Any other frequency between 2000 and 6500 kc. can be supplied. The output is thirty watts, 200 per cent modulated. Five type 6A5 tubes are used. The unit contains a volume control for adjusting the volume of side tone supplied from the microphone to the headphones. The price, exclusive of the plate, excluding a twelve-wire antenna, is \$250.

The Radio Laboratory of the Park Hill Airport at Denver, Colorado, announces a 3-lb., light-weight, "Shrove-Aero" transmitter for the private flyer, capable of delivering 15 or 35 watts depending on the power supply used. The three-frequency type (3185, 3120, and 6210 kc.) measures 8" by 8" by 7", and weighs 32 pounds. The price ranges from \$242.50 to \$425.45 depending on the number of crystal frequencies included. The dynamometer power supplies are mounted integral with the transmitter. Aluminum plate and other

accessories accessories, aircraft microphone, and antenna fibers, etc., can be supplied by the same company. The transmitter installation includes a side-tone panel and interphone system.

22-lb. Transmitter

Western Electric Exhibits Model at Chicago

A NEW LIGHT-WEIGHT TRANSMITTER exhibited for the first time at the Chicago Show, has been announced by Western Electric. The new model, designated as model 21A, weighs but 22 pounds including power unit, and measures 7 1/2" by 3" by 4 1/2". Two 6A5 tubes are employed, delivering 15 watts, crystal-controlled, to a properly matched antenna. The transmitter is intended for operation on the standard frequencies of 3185, 3120, and 6210 kc. but any frequency between 2,000 and 6,500 kc. is available for use, including the 42 airline frequencies.

A relay incorporated in the transmitter permits the same antenna to be used both for transmission and reception. The relay is controlled by a push-button installed in the emergency phase, which the operator "pushes-in-table." All accessory cables are supplied with the unit.

Other new Western Electric aircraft radio equipment includes a larger transmitter, the type 12CA, which covers both long and short waves, 250 to 550 kc. as well as 1,500 to 2,500 kc. The lower frequency band is useful in communicating with towers and coastal stations. The output, 50 watts, can be fed in a 250-foot feeding antenna, even at the lowest frequency.



Recent Transmitter of the Chicago Show shows the large and light new the flexible streamlined landing-type type is mounted on the top. Under the wire he carries the receiver, which transmits the landing between landing waves and the line of flight into a reading on the visual direction indicator.

not being made of a special landing coil for this purpose. The total weight including landing coil is 45 lb. and 9 lb. The crystal-mounted frequency control is within 0.025 per cent of the operating frequency, throughout the temperature range from -75 to +40 degrees Centigrade.

A hand-port filter, designed for use in connection with the non-humans radio range and weather broadcasts from the Bureau of Air Commerce stations, has also been made available by Western. A three-position switch may be converted to the filter to provide both range and weather, range alone or voice alone.



Western Electric 21A transmitter, 22 watts from 22 pounds.

KOLLSMAN

PRECISION AIRCRAFT INSTRUMENTS

KOLLSMAN INSTRUMENT COMPANY
INCORPORATED
1001 FORTY-FIFTH AVE., ELMHURST, N.Y.
MEMBER BOARD OF GENERAL AIR TITANIC, CLERMONT, CAL.

Buyers' Log Book

What's New in Accessories, Materials, Supplies, and Equipment



Criley Starter

An interesting new gadget for the light plane owner

HAVING THE DISTINGUISHING of being the first approved type under the light aircraft, the Criley hydro-pneumatic starter has already met with favor among many operators of Piper Cubs, Taylorcrafts, and other light planes. Although purely a manually operated starter, its principle, being actuated by the pilot through the medium of air pressure and oil, the Criley starter is a practical engine installation which should prove invaluable to operators of light airplanes, and which eliminates any necessity of climbing out of the cockpit to start the engine on any light plane. This is not only a feature of convenience, but also a marked safety factor.

Manufactured by the Criley Aircraft Industries of Downey, Calif., the Criley starter guarantees power to start the engine over from exactly through hydro-pneumatic action. Initial energy is applied by means of a piston actuated by a depressing lever moved by hand (the means of a crank on the instrument board, or adjacent to it. When the piston has been depressed, a lever is tripped and the stored energy is released, exerting a movable piston in the vacuum cylinder which actuates a rack against the starter gear to turn the engine until one full revolution. In case the engine does not start on the first turn this cycle may be repeated in less than a minute.

The Criley starter is not only of



Criley Starter

relatively low price, but it is of unusually low weight, only 6.5 lbs. complete. Also, most of the parts are of standard automotive manufacture, all of which are readily available for replacement. Since the air pressure energy stored for operation of the starting cycle reaches a pressure of approximately 150 lb. per sq. in. in the starting action to reach some extent this is possible in pulling the poppet through by hand, and it is claimed that this rapid turn-over provides relatively quick starts. In the event of a back-fire the starter merely recharges itself automatically and is ready to be tripped again without the necessity of hand charging.—*Aviation, March, 1935*

Do-All Contour Machine

A filing machine for the small shop

A NEW LOW PRICED MODEL of the Do-All contour machine has just been placed on the market by Continental Machine Specialties, Inc., Minneapolis. It is designated as the 500, due to its new 16" throat capacity and two stages of infinitely variable speed. This new model is rigid on the basis of the company's Mechanism. It incorporates many of the features of the larger machine, such as, automatic butt welder built-in, and four-way table tilt.

It is of heavy welded steel construction. While filing, the machine has a new two-piece backing member, which makes it unnecessary to remove anything for internal filing operation.

The range of speed is from 75 to 450 ft. per minute. The upper range is from 450 to 2,800 ft. per minute. Thus, shops may handle production sheet metal fabrication in this new machine tool, as well as Do-All and room work. As an inventory model, a Job Selector tells the operator speed to use for each job whether sawing, filing or polishing.—*Aviation, March, 1935*

Kinner Replacements

An-Park Success ATC on parts

DESIGN IN THE FIELD of airplane engine replacement parts is the action of An-Park, Inc., Glendale, Calif., in fitting a standard Kinner E-3 engine with a number of special replacement parts and running an official ATC fifty hour test under 16 of AC regulations. Among replacement parts appeared as a result of this test, but manufactured by An-Park, Inc., instead of the original manufacturers, are pistons, valves, cone section, cylinder heads, cone follower, piston, valve guides, etc. An-Park, Inc., operates its own machine shop in conjunction with a wide mail order price business operating throughout the United States, Canada, Mexico, Alaska, and South America. W. A. Kinnison, president of An-Park, reports a substantial increase in the demand for Kinner parts as a result of showing ATC ratings.—*Aviation, March, 1935*



Do-All Contour Machine

Brazing Alloy

Developed for use at unusually low temperatures

BECAUSE IT PERMITS BRAZING at unusually low temperatures, avoiding damage to structure of the vessels being joined, the new Easy Flo Brazing Alloy introduced by Hundy & Herman, (22 Fulton St., New York), is of special interest to the aviation industry. Easy Flo flows freely at 1175 deg. F. and works equally well on both ferrous and non-ferrous metals. It is being widely used for joining aluminum steel, steel steel, and many chrome nickel and copper-nickel alloys. It is also useful in joining dissimilar metals at low temperatures.—*Aviation, March, 1935*

"Grabit" Boat Hook

A handy accessory for airplane owners

A NEW INVENTION which has a distinct place in its plane equipment, has recently been placed on the market by the Thomas Langley Co., manufacturers of Portland, Maine. This new is known as the "Grabit" Boat Hook and is of greatest assistance in securing planes to mooring slings. A light line is splined into the eye of the boat hook and the other end made fast to the plane. When the hook engages the ring, it is locked in place by the tongue, shown in the illustration. The boat hook handle detaches itself from the hook and thus the plane is quickly released.

The "Grabit" Boat Hook has been approved by the United States Ministry. Its application in aviation was devel-



"Grabit" Boat Hook

oped by Mr. Glenn S. Payson, of the Thomas Langley Co., a licensed pilot of many years standing, formerly associated with Stevens, Inc., of Boston, Mass.—*Aviation, March, 1935*

Fills Tanks From Bottom

Valve now developed by Theo. L. Schenck

DEVELOPED by the Theo. L. Schenck Mfg. Co., 410 West Sixth St., Kansas City, Mo., manufacturers of aeronautical supplies, a new valve fills tanks from the bottom instead of the top as at present. It was found that like a gap is really seen to be at the side of the tank on second thought. Not only do airline operators want a great deal of time checking up on the wings of transport planes to fill the fuel tanks, but they also desire the fuel tanks some extent in their checking over it. Moreover, the personnel also realize that for maintenance records show that more than a score of men were last checked 1935 in full. Thus airplane wings while engaged in filling wing fuel tanks. And one of these men was permanently incapacitated by injury. Furthermore, with transport planes growing larger the problem of filling them will become acute. To offset this problem the Schenck valve has been developed under patents of the Fairbanks Co. of Los Angeles, Calif. The Schenck valve is a lock proof unit, in the bottom of the tank. To use it requires a special unloading unit on the end of the fueling hose, as well as a bottom lock compressor and a special vent.

The fuel pump system has been perfected by the Schenck Company and is now being supplied in use in several other operations. The fuel pump system on the whole side of the tank is actually lock proof, and shows the operator what quantity of fuel is in the tank being filled, then spending on the operation. The special vent is incorporated in the regular fuel filler cap, thus having the regular filler cap for use in filling the tanks of airplanes where the Schenck valve equipment is not used. The Schenck valve and makeup unit are for all practical purposes lock fastened and indelible. Through their use the tank may be filled or drained from the bottom, or the operator may draw off a fuel sample, all without disconnecting the makeup unit from the valve. The Schenck valve has been developed in two models to date, one having

a capacity of 150 gal. per minute at a fuel tank pressure of 20 lb. per sq. in., and the other a capacity of 250 gal. per min. at a pressure of 40 lb. per sq. in. Due to the use of a special spring and gravity valve in the Schenck valve unit it is possible to connect and disconnect the makeup unit without loss of fuel. Heavy rubber tires, or ball tires, are mounted on the makeup unit in order to prevent the operator or handle it freely, even roughly, without damage to the valve parts.—*Aviation, March, 1935*



Internal venting and spring loaded gate on the Schenck valve, shows valve body, spring, and cap



Schenck makeup unit and valve shown coupled

Chargeurs Réunis Takes Another Sikorsky



LAST year Chargeurs Réunis, famous French steamship line...through its newly-formed subsidiary Aero Maritime...opened a 3,000 mile *air* service along the west coast of Africa between Dakar and Pointe Noire...selecting for this operation a fleet of three Sikorsky amphibians. These Sikorskys have performed so brilliantly that increasing patronage has required additional equipment...Another S-43 is *on* the way.



SIKORSKY AIRCRAFT

BRIDGEPORT, CONNECTICUT

ONE OF THE FOUR DIVISIONS OF UNITED AIRCRAFT CORPORATION

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THE AVIATION

NEWS

VIEW, COMMENT, FORECAST

DAVID SAYRE
C. F. McLaughlin, *Aviation* Editor
E. B. Larkin, *Aviation* Editor

MARCH 1938

U.S. Industry Breaks All Records

(Story on page 52)

FLYWAY HASTENED: The Wagner Crusader (P. & W. design and H. B. Wright) head up for a delivery flight to Mexico City. Flown by Republic Army pilots under the command of Capt. Karl M. Wagner, the flight sets a precedent as the first passage of the U.S. of an entire squadron destined for a foreign government. Ships were made at Washington, Cincinnati, New York, Newark, San Antonio, Omaha (Mo.) and Mexico City. The ships are specially designed for high altitude operations.



With World

ONE CENT A YEAR: Carroll Corp., member of Pan American Airways Division (PAC) since 1935, has sold 10,000 shares per year for a U.S. Airways air mail contract with Harbor Branch (PAC). P.O. Pan Am, get the contract. No one else turned in a bid against.

A GREATER TEMPELHOFF: Germany is rushing a huge program of improvements on its Berlin terminal. Runways will be 1,000 to 1,200 ft. Photo below shows extent of magnificent rail-road buildings reached through underground roads and railways. Plans laid directly under a big, multi-story roof to land and unload.



AVIATION
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COVERS THE AIR FRONT: Devo Frenck of the Associated Press has been awarded \$250 and TWA's trophy for "absolutely deciphering the best informed writing" on 1937 aviation efforts.



25,000 HP. SMILE: Al Wenzon before his new 50 hp. unit. Aeronca has ordered 500. (See page 30 for details)



SH2, SH14, SH20: Falar, John and Geoffrey De Havilland, sons of Capt. Geoffrey De Havilland. All are on the De Havilland Aircraft Company staff.



FRANK LEGG: The industry lost one of its finest engineers and most splendid personalities when Geoff Vultee crashed in Arizona on Jan. 25, while flying his Vultee through a blinding squallstorm. Frank met Lucius Rust, Vultee, of 35 years of age, had obtained top rank among American aircraft designers. A graduate of Caltech, he was originally associated with Atlas Lockheed and John K. Mooring in the development of the Lockheed "Vega." He served for some time as chief engineer of the Lockheed Aircraft Corp., and about five years ago established his own firm, developing the Vultee aircraft development program. The plane was manufactured by the Vultee Division of Aviation Manufacturing Corp., which later established itself in a factory at Downey, Calif. There the current Vultee stockholders were placed in production.



MERV V. C. M. Moular has left Brewitt to be vice president of American Airlines.

AVIATION PEOPLE

Who's Who in This Month's News



EXECUTIVE SHIFT: From a long time association with the Curtiss-Wright Corp., initially as assistant to the president, Albert L. Luchins (right) goes to Aviation (left) Corp. as executive vice president. He will serve under President Rust (shown) in the general supervision of the reorganized corporation's several manufacturing subsidiaries. His resignation from Curtiss-Wright was effective May 1.



PUBLIC RELATIONS: To become manager of public relations for Curtiss-Wright, Ronald E. Galt will shift the focus of his activities from engineers, whom he has had charge of publicity (including the editorship of "Trade Wind") for Wright Aircraft Co., to G-W headquarters on the New York side of the Hudson River.

NEW PRESIDENT: Elected to head up the reorganized Aviation Manufacturing Corp. is W. H. Rust (left). Formerly president of Locomotive and more recently chairman of the board of New York Shipbuilding Corp., his activities will cover the Vultee Division, the Avian Division and the Licensing Engine Division. He served as a Captain of Ordnance during the war.



MAN AT WORK: Stanley Bonica, former American racing pilot, holds the swiftest of his new sail plane, the plane to market it at \$700 some place; \$400 in kit form. It contains an aircraft, airplane and contains an aircraft.



COUNTING THE HOUR: James H. Butler, Jr., Jack Vian, and Maynard W. Robinson, representatives of Chicago Air Race, get Valentine greetings in the form of the first recording. Reflected in their faces is a satisfactory report.

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Solving Space Structures Graphically

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solved in the same manner and the loads in F and JF are found to be of the same magnitude as those in I and D , but of opposite sign.

The resultant load at (3) could be found by laying off the stresses in J and F , but since the latter two are numerically equal, the horizontal components (left and right components) are used to balance each other, so that only the vertical components need be considered. This gives R , and the determination of the stresses in K and K' is made in the usual manner followed by the marking and by the transfer of the joints (11) and (12).

This completes the solution of all the members of the front bay of the space truss structure.

An examination of the joints (11) and (12) shows that there are too many members of each, so that no need find solve joints (5) and (6), each of which has only three members (the others are now applied stresses).

One must be cautioned in the determination of the applied resultant load at the screen joints. It is best, starting with one view, say the top view, to take care of all the stresses in that view first and then transfer the top projections of the stresses in the side view members which end at the particular joint in question.

Thus, at joint (12) the stress in JF is taken as the first applied load, as a line three at end and parallel to JF is laid off the stress in JF . Then the new point is given a line parallel to G and the stress in G is laid off as it. This takes care of the loads in the top members. There remains the load at the side member W (See schematic sketch). Slide the top projection of K vertically with A' , and project the stress in K' on A' , and drawing a line parallel to A' , lay off all at the but and project the magnitude of the top projection of the stress in K' . The resultant R is a vector drawn from joint (12) to the last end point and pointing toward the letter.

The pressure of any applied external load at joint (12) would not complicate matters much. It would only require one more vector parallel to the applied load and equal to it in magnitude.

The same procedure is followed now in determining the side projection of the resultant load at (9), hence the stress in W is taken as the first applied load; the stresses in G , BF and A' are

then projected on A' (in side view) and their projections (or their vector sum, since they are all collinear) are drawn at the side point, thus determining the resultant R . (In this case it happens that this view is seen; that is, however, merely coincidental.) As a check on the correctness of the work, the last end point in the side view must lie exactly below the same point in the top view. By inspection joint (13) is found to represent case 4b. Therefore, either view may be solved independently of the other. This gives the stresses in F , G and K .

A WORD OF CAUTION with regard to the determination of the resultant R , to ground the applied loads may be drawn either with their full end and as with their feet and touching the joint. However, the first applied load has been drawn with its tail end at the joint and the other loads must follow in the order indicated before and the resultant R will have its tail end at the joint and its head end at the head end of the last vector. It is, however, the first applied load vector has been drawn with its head end at the joint, the resultant will have its head end at the joint and its tail end at the head end of the last applied load vector.

Joint (10) is also one of the resultant is determined as for (9) and the method of solution is the same. The stresses in M and JF are the applied loads at joint (10). These two are then given the resultant R . Carefully applying joint (10) cannot be solved since it consists of four members. However, suspicion about that the two crosshairs members load to joints (11) and (12) while M and JF are the other two. It appears, from the supports are assumed to be rigid while the joint (11) and (12) defect, and therefore, since the resultant R is in the plane of M and JF , it may be assumed that it is taken entirely by the latter two members.

Turning to joint (11) and (12), we find that each of these contains five members. Since the maximum number of members we can solve for is three, this is a statically indeterminate structure. Solving L at the intermediate member, we proceed to solve the so-called determinate structure.

It consists of two triangles, one with an apex at (11) and the other at

(12). Since neither view of these triangles has any superfluous members, they represent the general case and are solved by the general method.

Starting with joint (11) we lay off in side view the stresses in G and G' and then the projection on D (in side view) of the stresses in D and A' . This gives the side view of R , in the bottom view the resultant is found by first summing for the stresses in D and A' next for the projection on D of the stress in G . After determining the line vector of the applied resultant load, we name member S the R member, W -the M member and L -the JF member. This is carried out in both views. The graphics given for the solution of the general case is now followed exactly and can be identified by means of the letters designating the various points. The resultant stresses are solved all at the supports and are marked correspondingly. The points marking the magnitude of the stresses in the side view must be exactly in line (vertically) with the corresponding points in the bottom plan view. (Note that the cross-diagonals S and J' have points in both the top and the bottom views. Their projections on the bottom view were used in conjunction with their side view to determine the stresses in them.)

Joint (12) is solved in the same way as joint (11).

The next operation consists in walking the stresses in all the members. The upper and lower trusses in the first bay are actually in true view (their side projection is almost horizontal).

Hence the stresses in these members may be solved directly in the top and bottom plan view; members B and C , also B' and C' have quite an appreciable angle with the horizontal in end view. Hence, the stress vectors have to be resolved first, in order to find their true length. In the second bay members Q , Q' , N and P are in the plane of the paper. The stresses are solved directly. T , T' , M and M' are practically in the plane of the paper in the top view and may be solved in this view. It will be found that small calculations at members have practically no effect on their true length (from $L' = .995$, on $T' = .994$, on $T' = .994$, on only 1/1000 less than the true length). Members S and J' are almost horizontal in the bottom plan view, hence their stresses may be solved in side view. S and J' are one of the horizontal in both views. They must be resolved before the stresses can be solved.

(To be continued)

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Operators' Corner



An exchange of ideas on the problems of the commercial aviation industry

QUESTION: On what has been your worst loss in selling airplanes to operators who have submitted the insurance? Have you lost the money and an extra pilot position of value returned? What advice have you used to increase the likelihood of selling these airplanes?

Expert Opinions

As CHAIRMAN of THE AVIATION COMMITTEE of the National Society of America, and in that capacity a member of the Bureau of Air Commerce Advisory Board, I was assigned in the discussion of your Question 27 "Does Life Insurance Payoff Sales Representatives?" in your December number.

I have found in many instances that an understandable attitude of operators was toward life insurance is due to misunderstanding. In general, a life insurance contract effected as a time when the applicant had no present, past, or intended connection with aviation other than as an occasional passenger is considered as an adequate figure. The proportion of policies which are returned because of such connection is very small. This is pointed out by your correspondent, Mr. Alfred H. Brown.

I have sometimes suggested that the insurance argument is used to end of the release of airplanes or airplane travel and that the prospects in question would not receive other means of this one did not work.

It is true that a prospective airplane purchaser who expects an increase in holding of his airplane will have to pay a higher price if he buys it after becoming a pilot. This is unavoidable even if airlines become much less frequent in relation to the amount of flying, or after the time when aviation operators' plane is nearly as common as driving one's own car. That is, the coming situation is due to the fact that the choice of having to pay a claim under a given period of time is much greater in the case of the average man who is new a pilot than in the case of the average man who

is not. It is true that some new pilots will become pilots and we have to consider these insurance as the outside risk, but they are the complete and probably will be for some time to come.

It would be easier for us if we did have to give up much consideration to the fact that one man flies a great deal more than the fact that another man is a pilot of an airplane, but with the fact as they are we can so more do that than we can ignore the fact that a man of fifty is more likely to die in the next year than a man of thirty.

Your correspondent mentions two problems, policies which carry riders and the cost of unscheduled policies. The disease nearly for the former is to buy full coverage insurance at the appropriate price from one of the insurance companies which sell it. However, a new pilot may find difficulty in obtaining full coverage insurance at any price and he has passed the non-renewable stage that so many pilots go through after being certificated. The other problem, that of a fair price for full coverage insurance on pilots, is a difficult one to express private pilots, since they differ so much both in ability and in amount of annual flying time (or so much in the winter). One of your correspondents speaks of a fair price as "a low reasonable figure." I am afraid that at the present time a reasonable figure and what many pilots would regard as a low figure are not quite the same thing.

The reason of insuring private pilots is one on which the law would not be based and consequently there is your reader, more they will.

Next Month's Questions

QUESTION 28: When specifically include there you want to increase your business? What time of the month have you used and in what time has been discussed? Have you made use of insurance companies' advertising that will help you? What related methods have you developed for this purpose?

desired one side of the problem, can offer some general suggestions.—J. R. Housley, The Traveler Insurance Company, Hartford, Conn.

QUESTION 29: What are your activities in the new Civil Air Regulation discussed by the last issue in the December issue of AVIATION? Are the requirements existing toward insurance, qualifications for membership, membership conditions, etc., in the best interests of the flying public? Are there any A.A.A. portions in which you feel?

No Fear!

As A PRIVATE PILOT, business pilot and the owner of a Stinson Taper, I am an owner of practically all of the new Civil Air Regulations, except that portion pertaining to my taking a license, which is a private pilot, riding in my ship with dual controls required.

Here in California, we have a good deal of "no fear" to be sure as well as much flat country that requires the attention of a navigator as well as someone operating the controls on various trips. Whenever we go fishing, hunting, or on a business trip, we cannot go together and alternate the flying unless one of us possesses an instructor's rating.

In my short years of flying, about eight years, I have taken hundreds of people on their first flight. A good percentage have purchased planes, or now use the air line as charter sight-seeing planes, or are taking up flying themselves, or late, but not last year, "flying aviators in general" instead of being ruled against it, if that person of the C. A. R. that I mentioned above is not intended, it will be my realization, work quite a hardship on the manufacturers of private ships. A large percentage of the reason for owning a ship and the fun of running it is just the C. A. R. pertaining to what I mentioned above is not changed.—F. L. McMillen, President, Guilford Company, Guilford, Conn.

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similar records since—*ponder!* Gasoline gave aviation its first practical power-weight ratio—a ratio that lifted a heavier-than-air craft off the ground and sustained it in flight.

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Pioneering in the Philippines— STEARMAN TRAINERS



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